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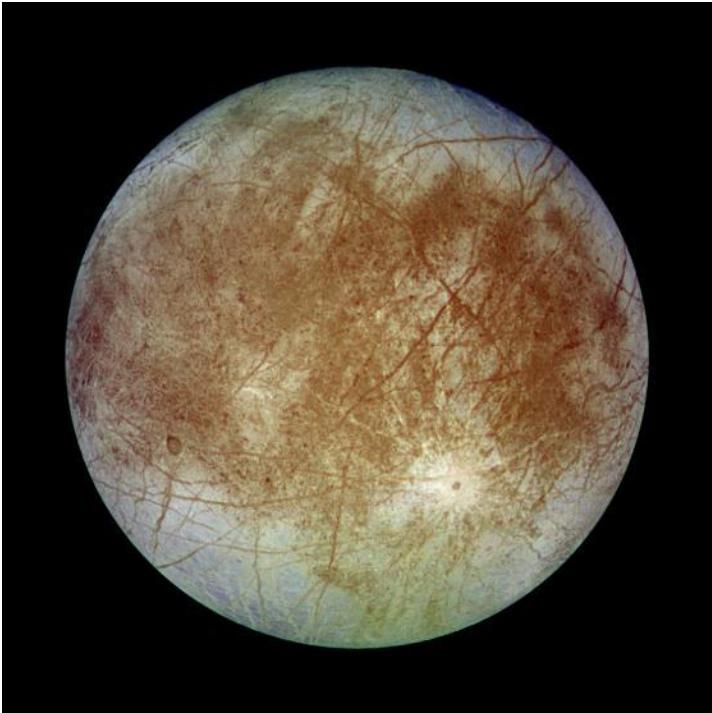
Callisto and Europa: a simulation study for gravity field determination from orbit tracking data

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PSD.1: Satellite dynamics: new developments and challenged for Earth and solar system sciences

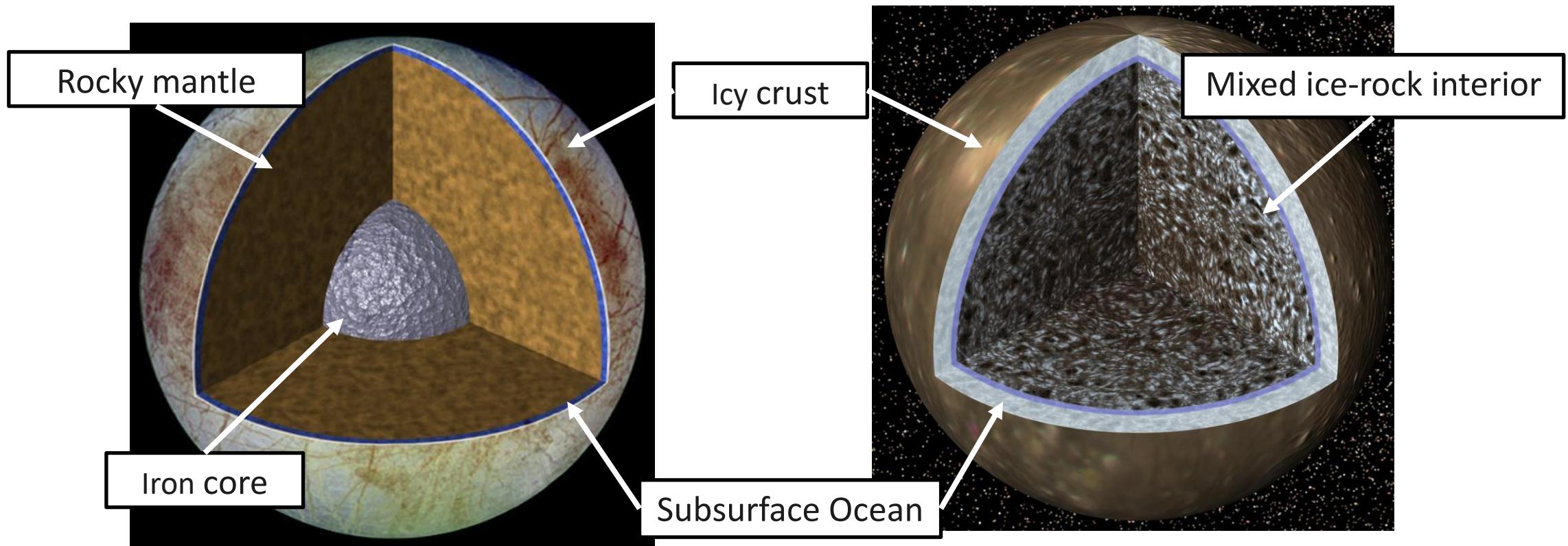
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(3) *NASA Goddard Space Flight Center*, (4) *IRAP, CNRS-Université Paul Sabatier, Toulouse, France*,
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Europa & Callisto



Radius	1561 km	2410 km
Mass	$0,008 M_{\text{Earth}}$	$0,018 M_{\text{Earth}}$
Distance to Jupiter	670 900 km	1 882 700 km
Orbital period	3,551 days	16,689 days
Surface	Cracks and streaks	Heavily cratered

Europa & Callisto: internal structure



Evidence of water plumes on Europa

Planned missions to the Jovian moons

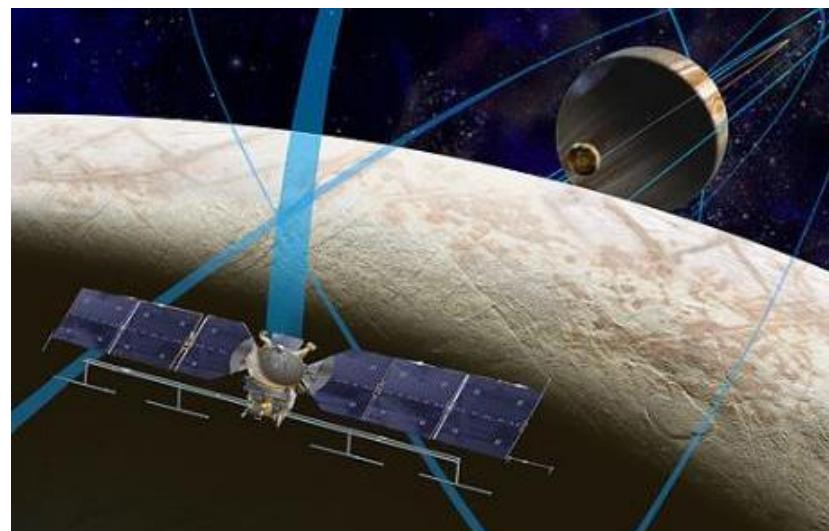
- **JUICE (ESA):**

- Launch: 2023
- Jupiter arrival: 2031
- 2 Europa flybys
- 21 Callisto flybys
- Orbit around Ganymede

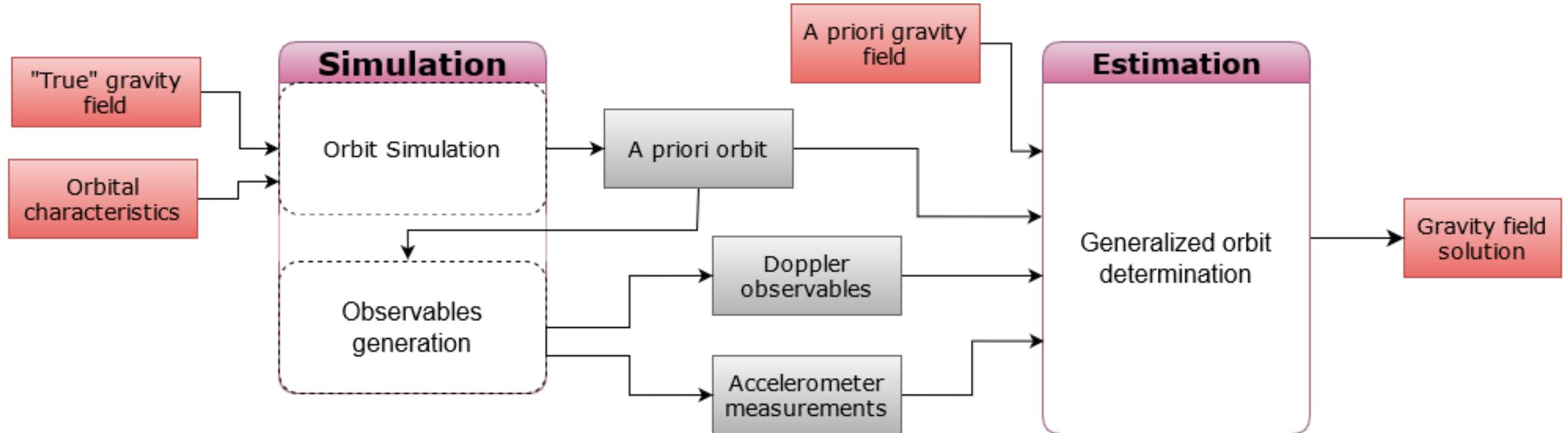


- **Europa Clipper (NASA):**

- Launch: 2024
- Jupiter arrival: 2030
- 45 flybys of Europa



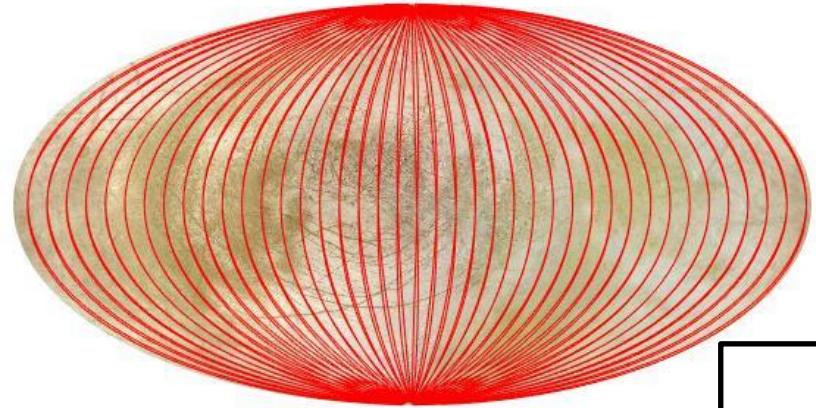
Our simulation pipeline



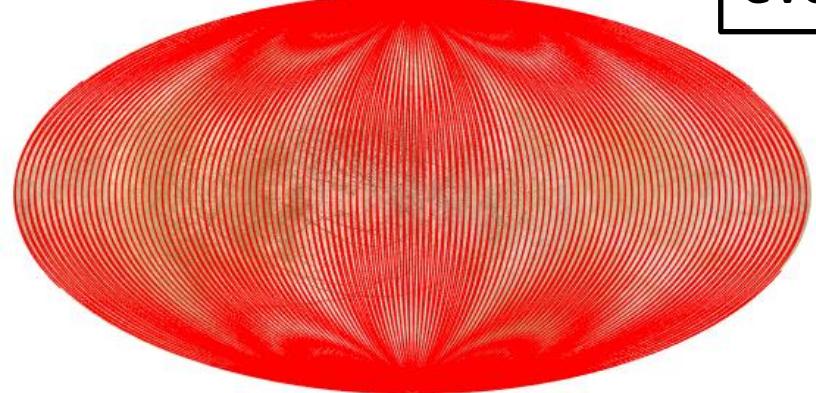
- Orbit propagations and the whole gravity field recovery process were based on a development version of the Bernese GNSS Software.
- 3 months mission
- 2-way Doppler:
 - Fixed noise: $\sigma (\tau=60s) = 0.1 \text{ mm/s}$
 - Detailed noise model (incl. solar plasma)

Repetitive Ground Track Orbits (RGTO)

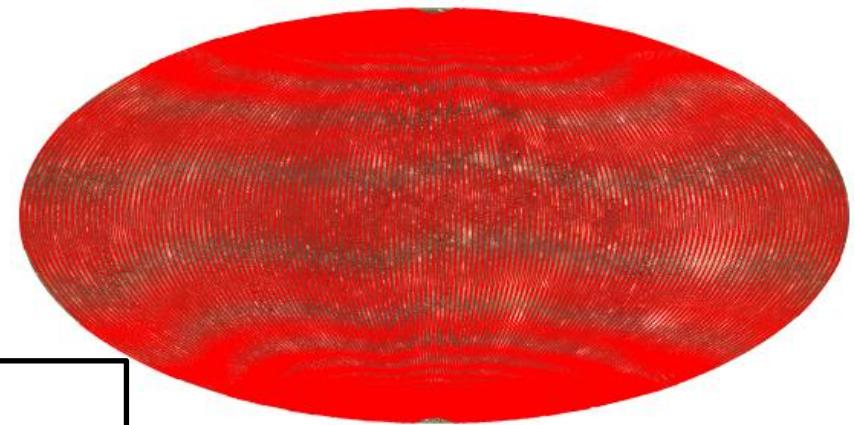
1:41 Europa RGTO



3:119 Europa RGTO

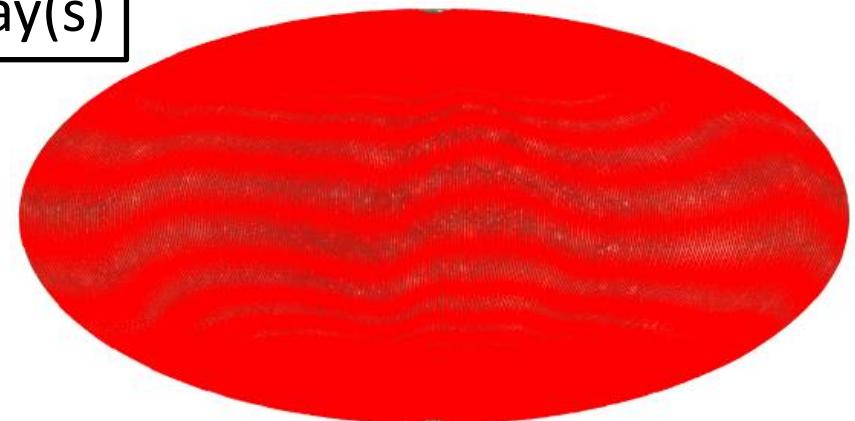


1:146 Callisto RGTO

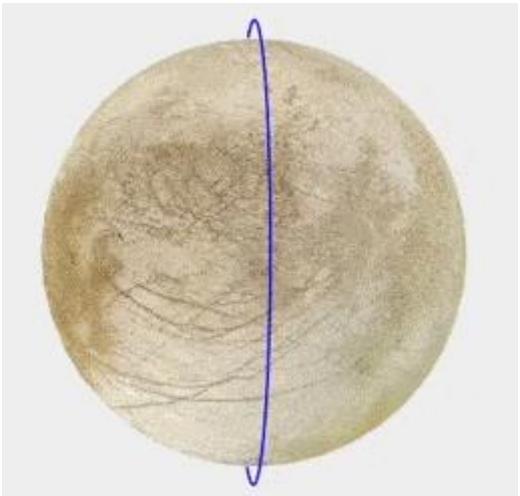


m:R RGTO:
R orbit ground tracks repeat
every **m** Europa/Callisto day(s)

2:293 Callisto RGTO



β_{Earth} angle

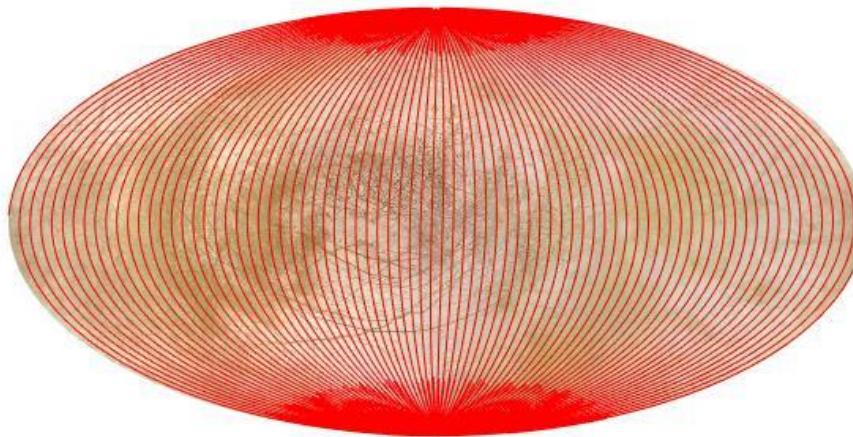


• Earth direction



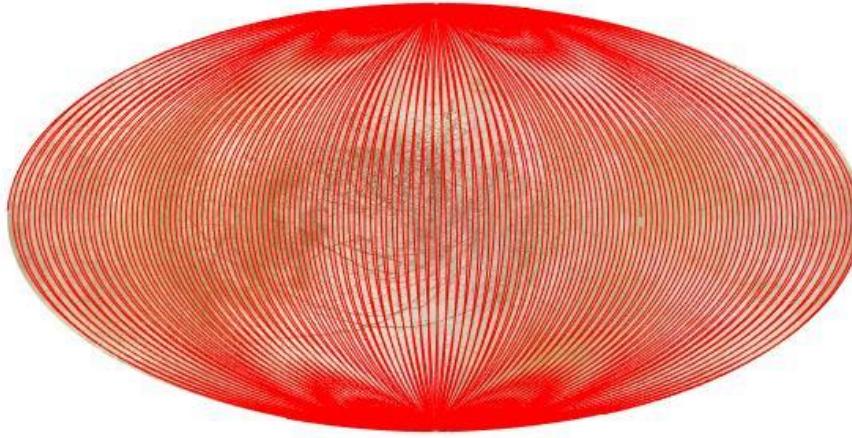
Edge-on orbit

$$\beta_{\text{Earth}} = 0^\circ$$

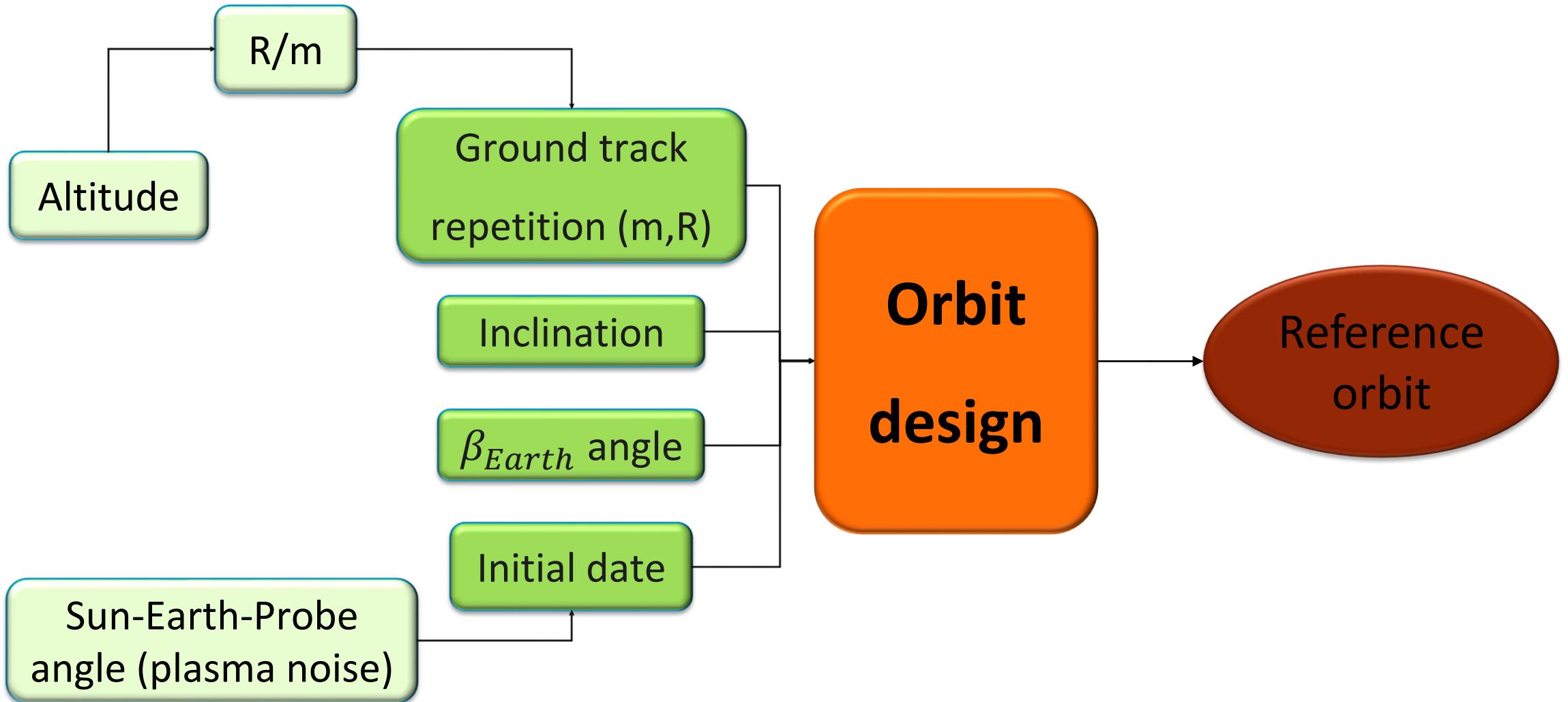


Face-on orbit

$$\beta_{\text{Earth}} = 90^\circ$$



Orbit design toolbox



Force model and synthetic gravity field

Force model:

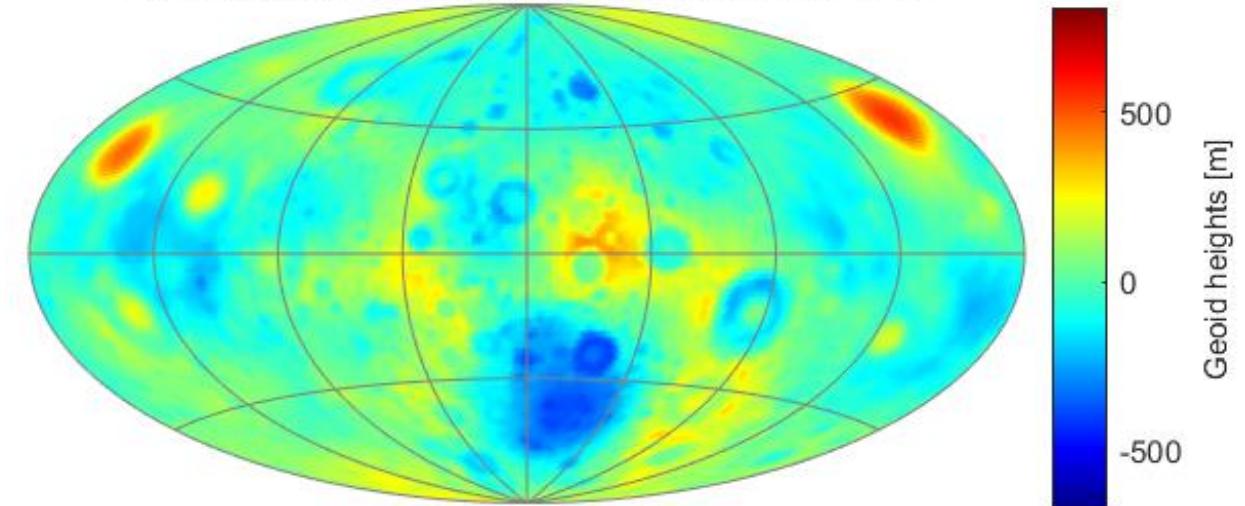
- **Callisto/Europa:**
 - Synthetic gravity field
 - Tides ($k_2=0.3/0.257$)
- **Jupiter:**
 - Point mass
 - Zonal coefficient (J2 to J6)
- **Other 3rd body:**
 - Other Galilean moons
 - Sun
 - Other planets
- **Non gravitational acc. (NGA):**
 - Direct Solar radiation pressure (SRP)
 - Planetary radiation pressure (PRP)

Synthetic gravity field:

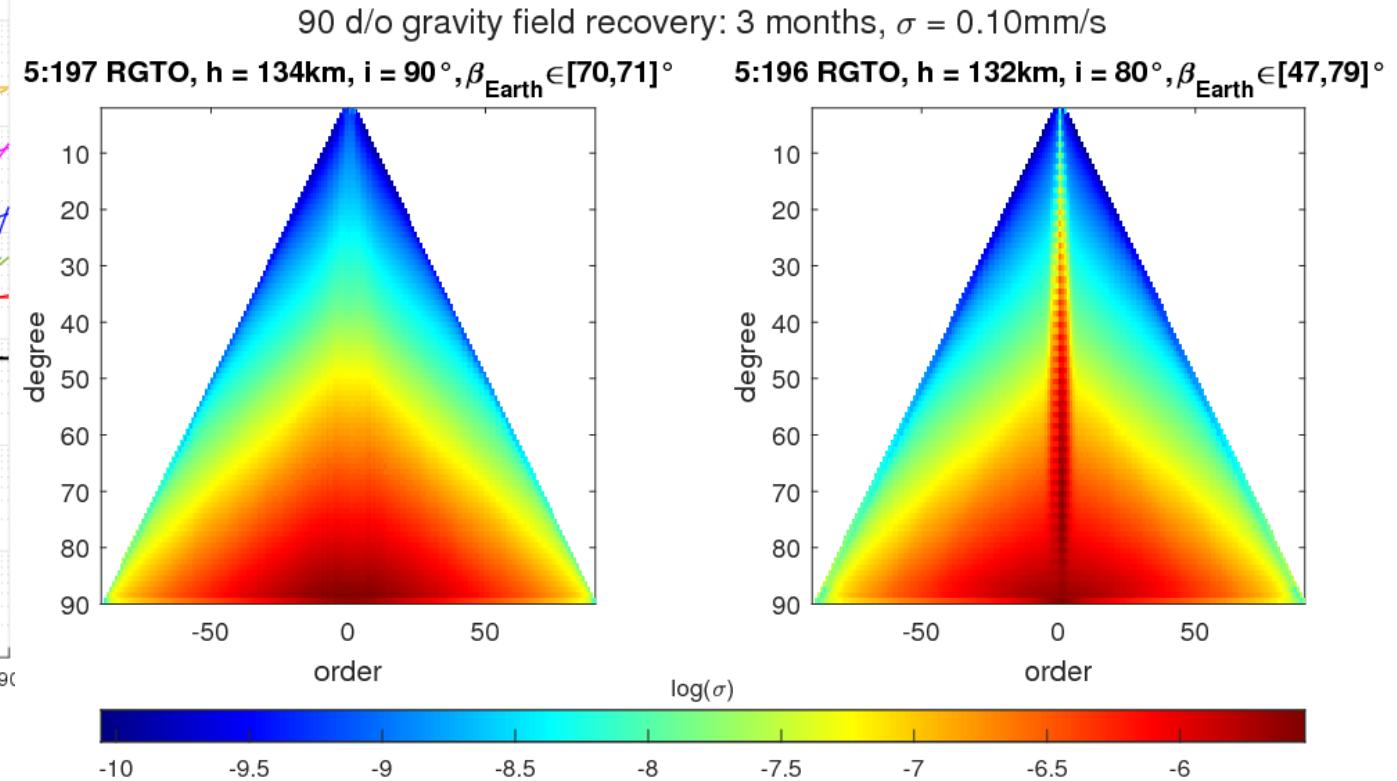
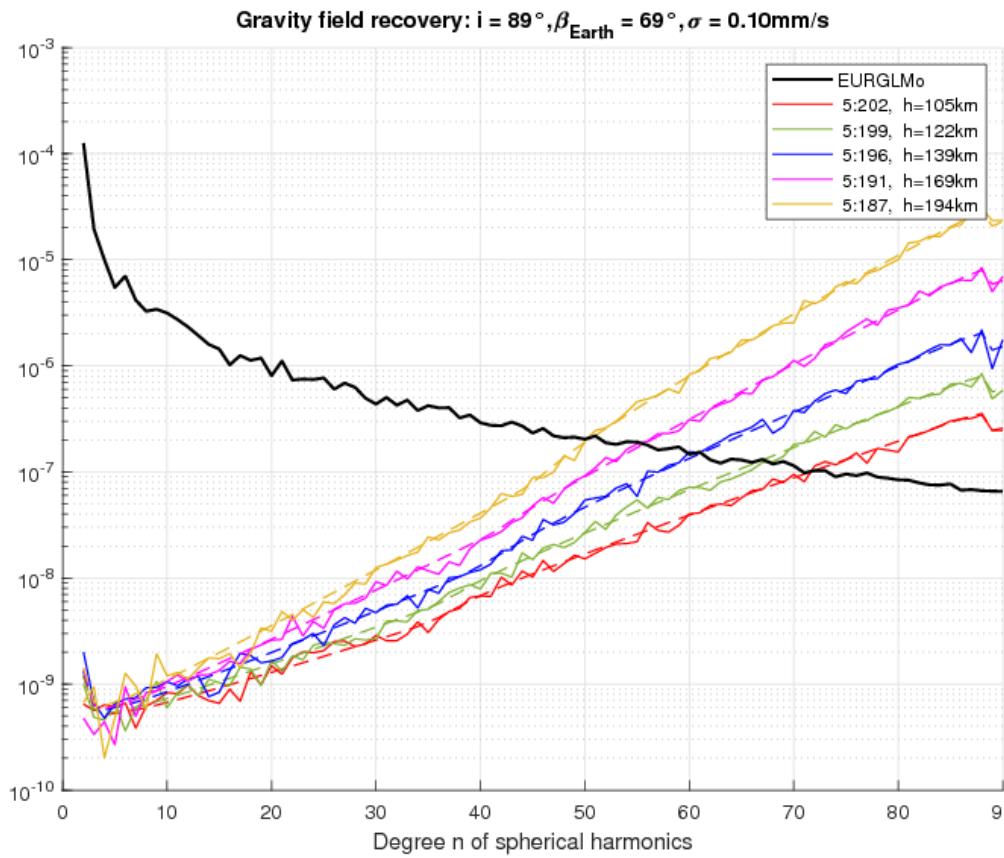
$$V(r, \lambda, \phi) = \frac{GM}{r} \sum_{n=2}^{n_{max}} \sum_{m=0}^n \left(\frac{R_e}{r}\right)^n P_{nm}(\sin\phi) (C_{nm} \cos m\lambda + S_{nm} \sin m\lambda)$$

- Up to degree and order 2: from Galileo mission
- From d/o 3 to 100: Scaled Moon's gravity field

Europa gravity field: EURGLMo (d/o 3 to 100)



Influence of the altitude and inclination

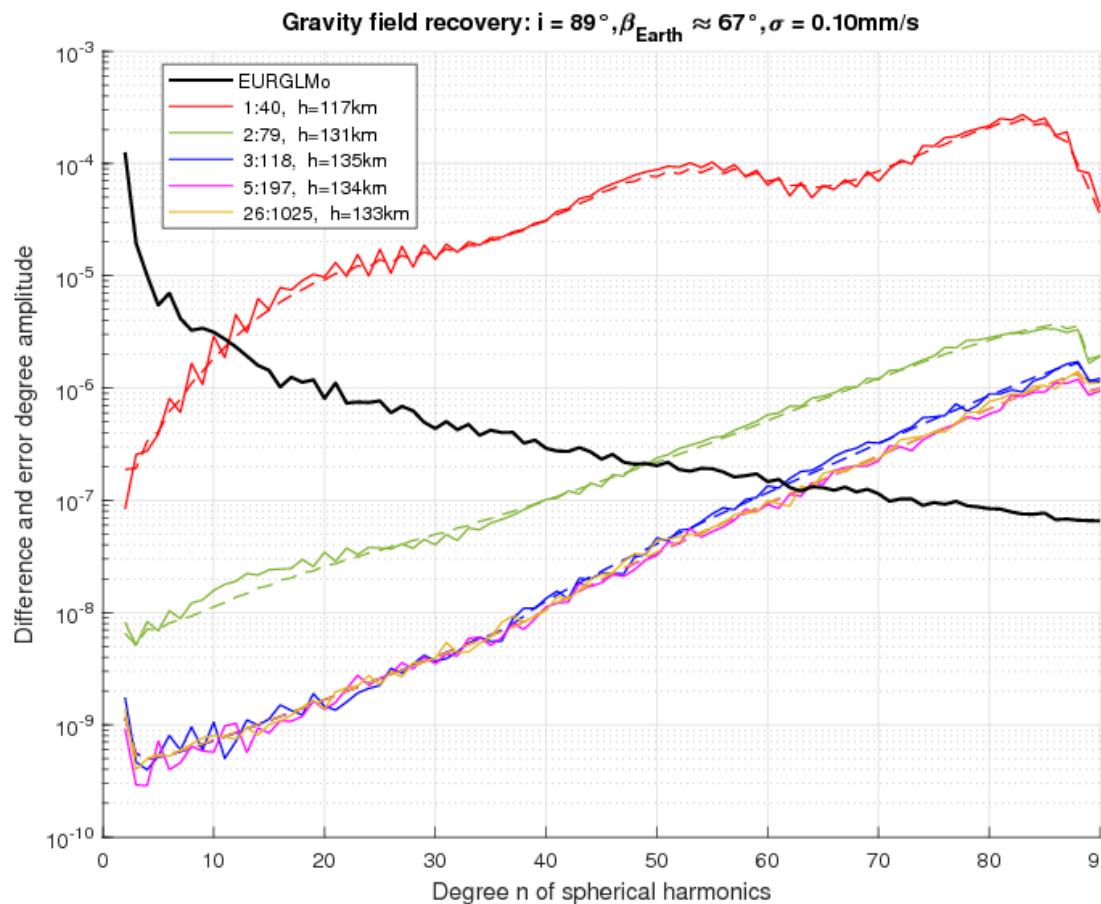


$$\Delta_n = \sqrt{\frac{1}{2n+1} \sum_{m=2}^n (\Delta \bar{C}_{nm}^2 + \Delta \bar{S}_{nm}^2)}$$

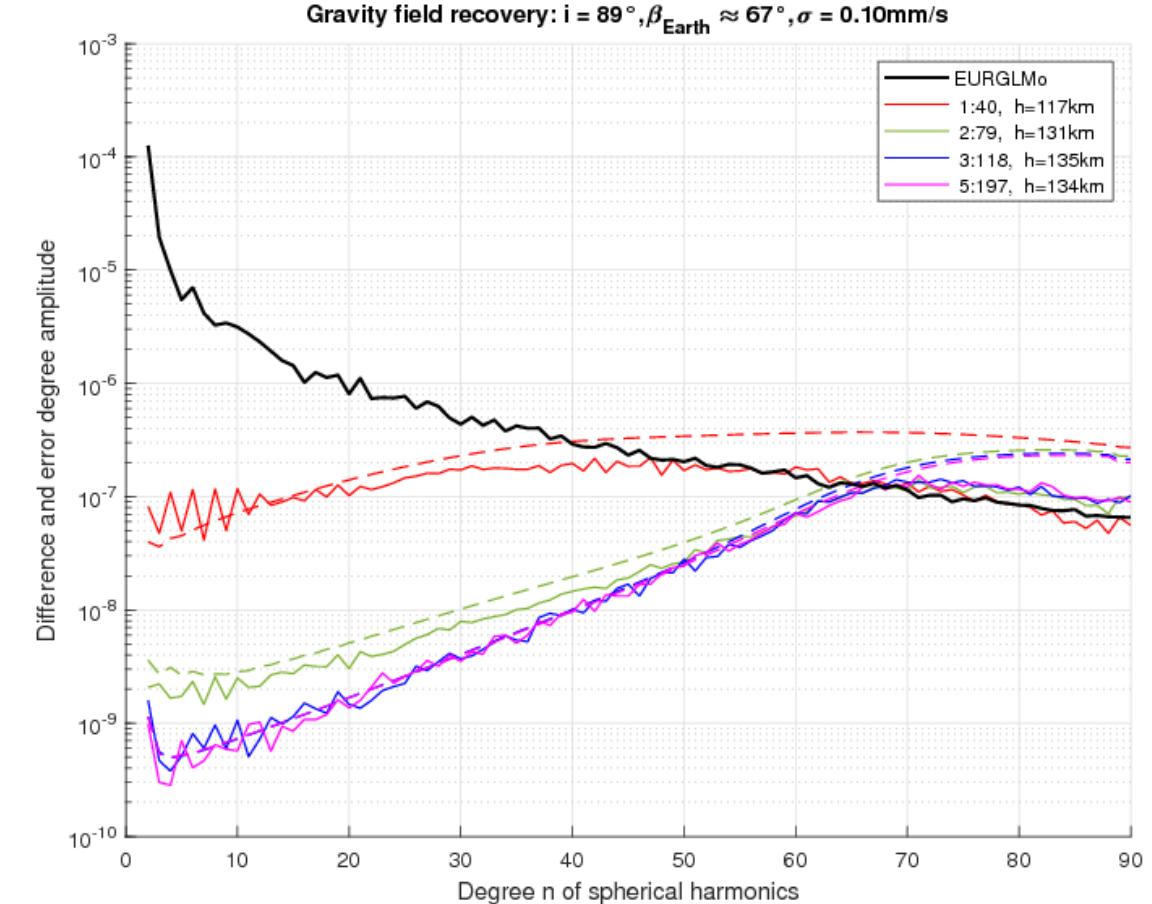
Influence of ground track repetition

$$h \approx 140\text{km}, i = 89^\circ, \beta_{\text{Earth}} = 66^\circ$$

Free solution

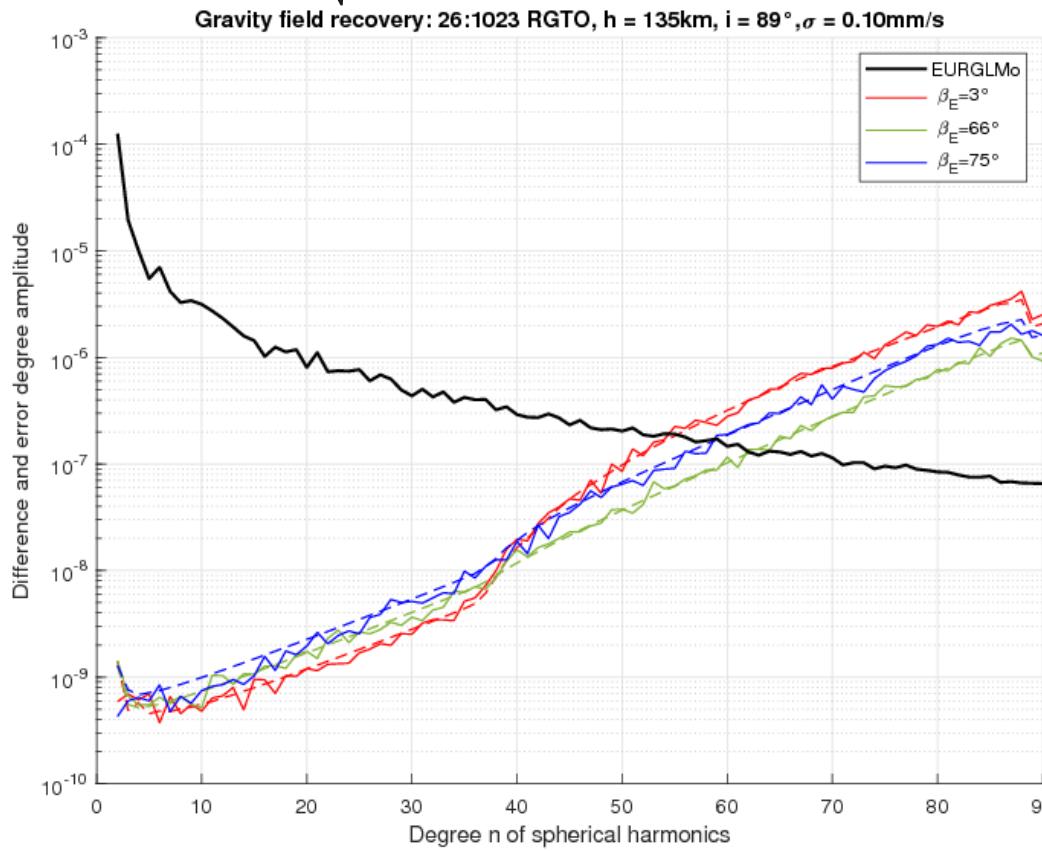


Kaula regularisation

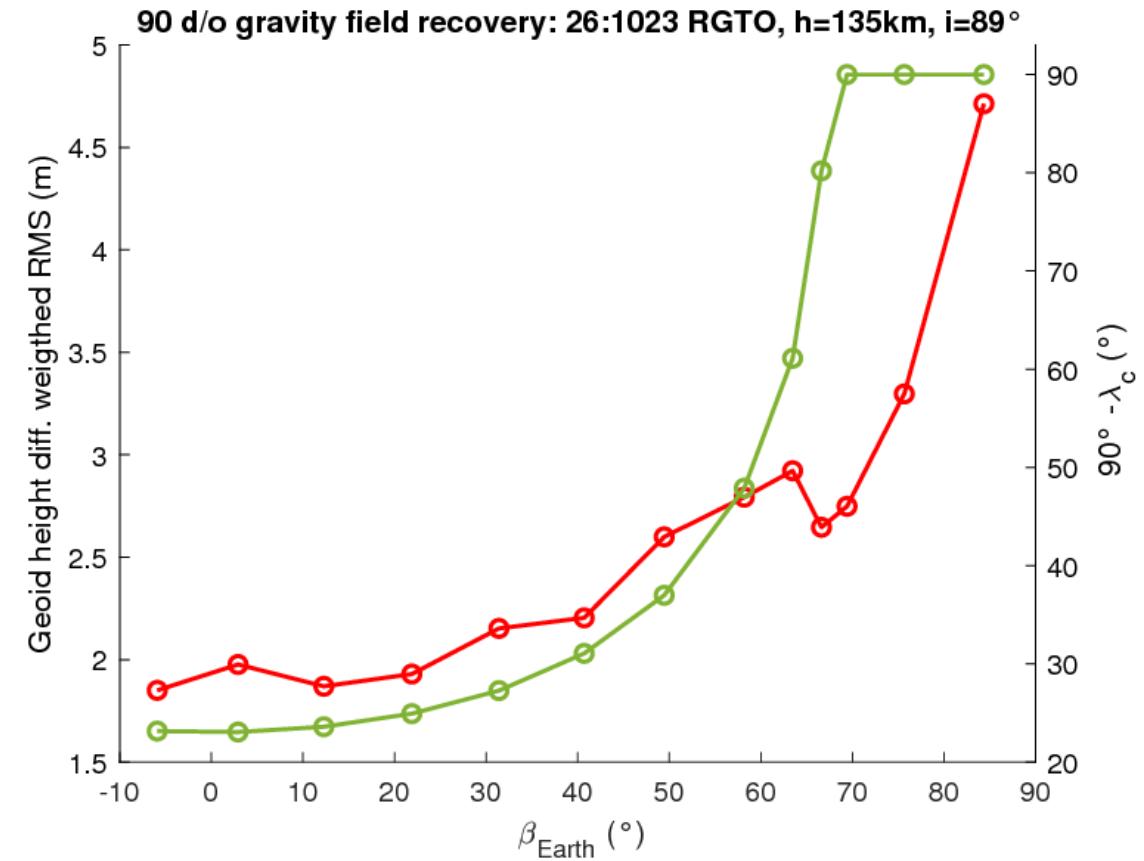


Influence of Earth beta angle

$$\Delta_n = \sqrt{\frac{1}{2n+1} \sum_{m=2}^n (\Delta \bar{C}_{nm}^2 + \Delta \bar{S}_{nm}^2)}$$

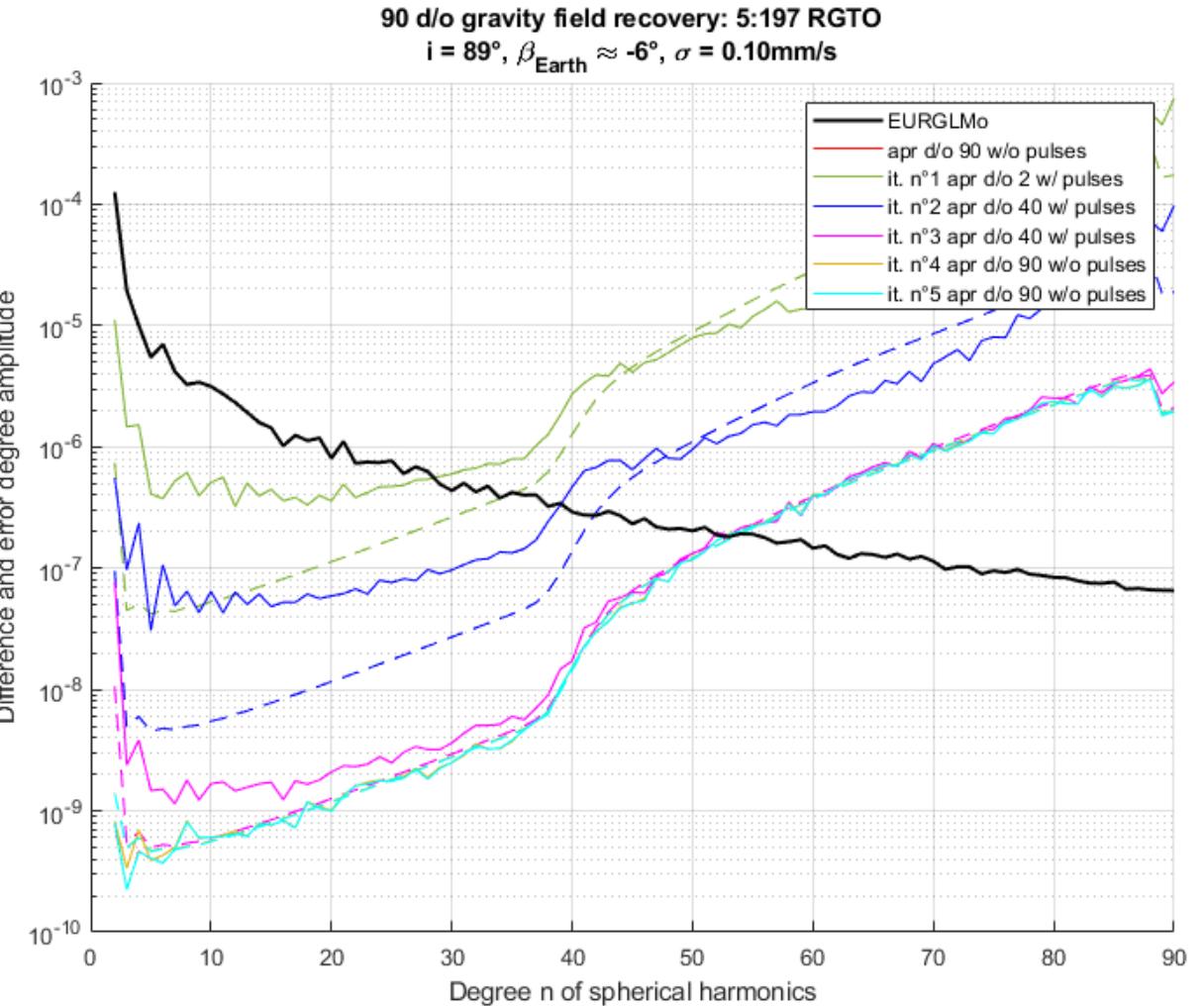


$$(\Delta g)_{WRMS} = \sqrt{\frac{\sum_{\theta,\phi} \cos(\theta) \Delta g_{\theta,\phi}^2}{\sum_{\theta,\phi} \cos(\theta)}}$$



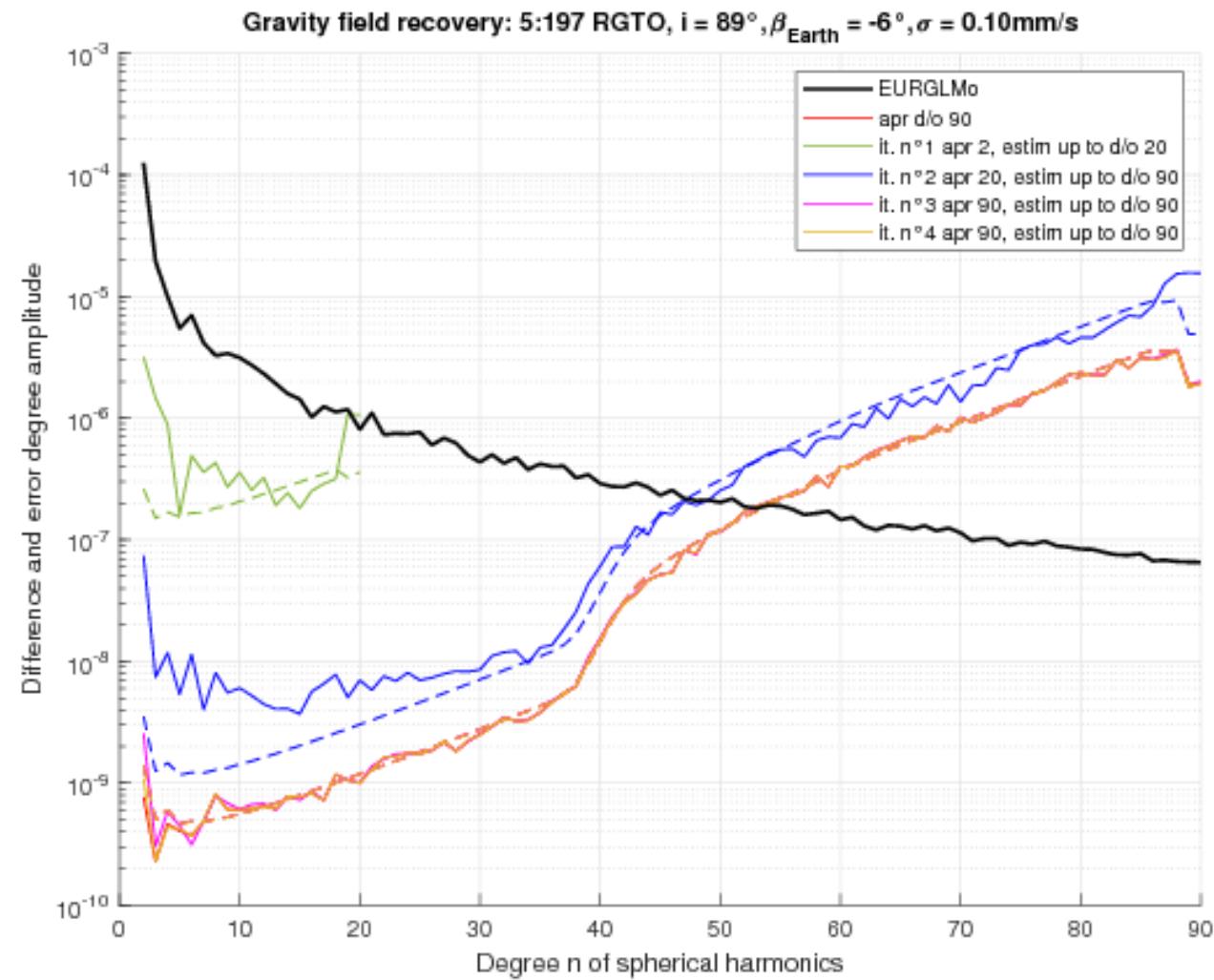
Using a degree-2 a priori gravity field: making use of pulses

- Starting from the truth, truncated d/o 2
- Pseudo-stochastic pulses
(instantaneous velocity change) every 60min to help orbit convergence
- 1 m/s constraints in all direction (Radial, Along-track, Cross-track)
- Affects the low degree
- Last iteration w/o pulses



Using a degree-2 a priori gravity field: co-estimation of the low degrees

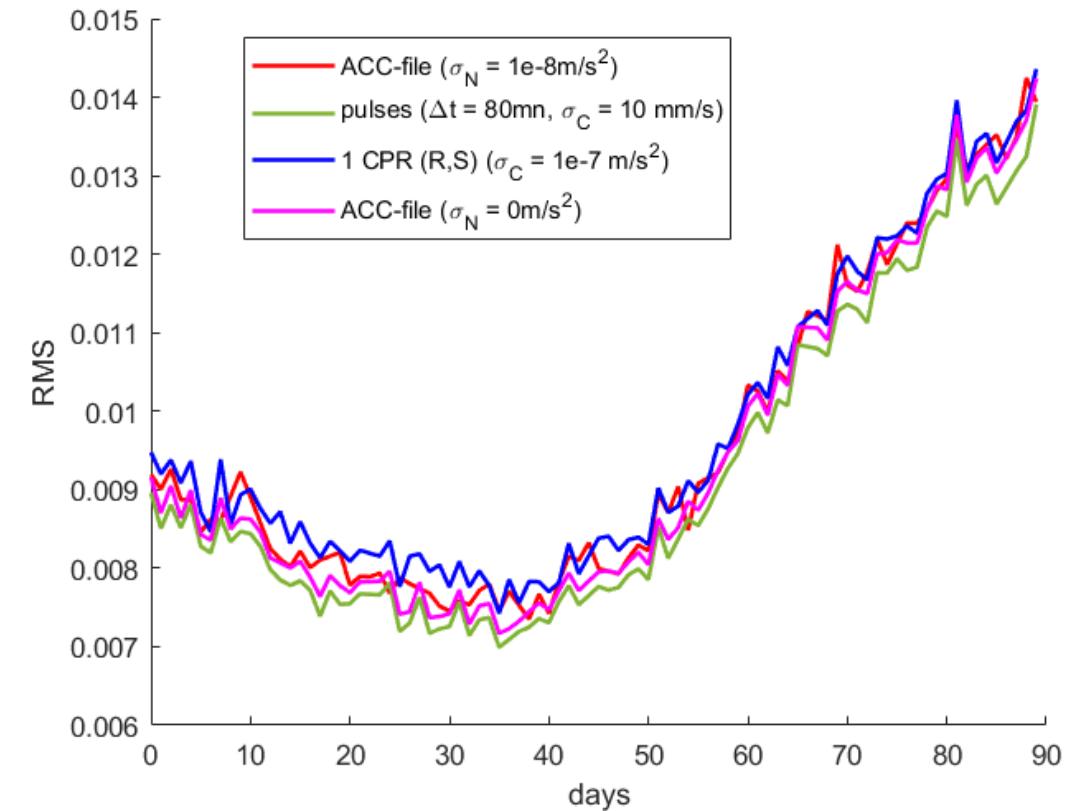
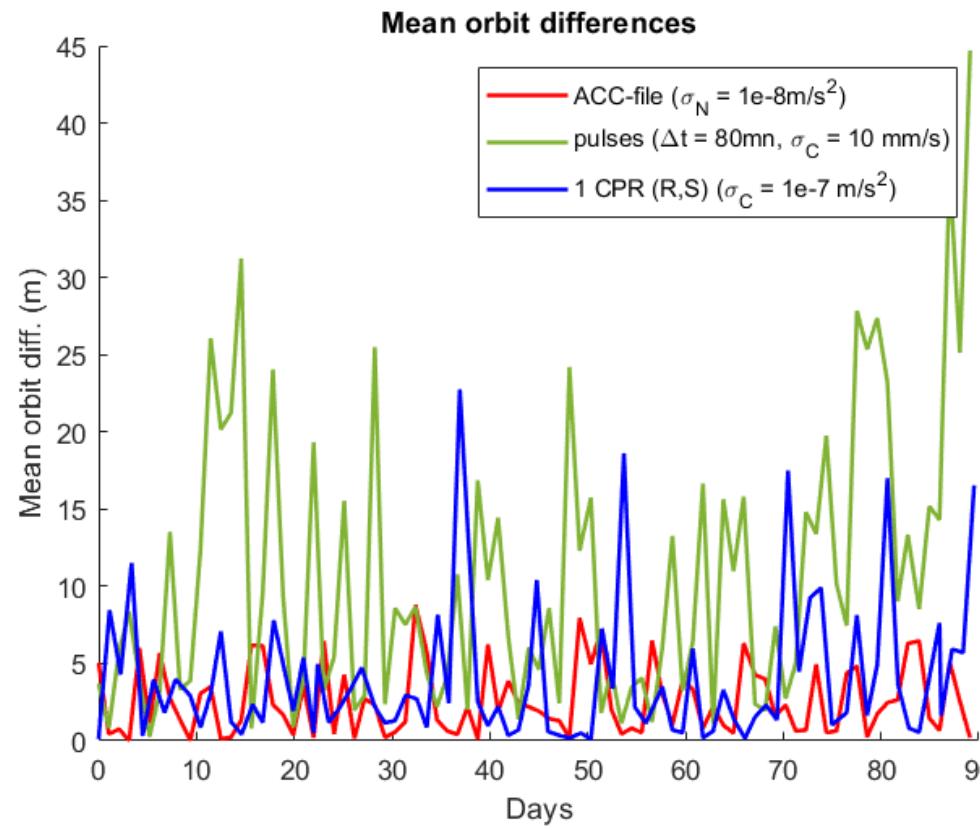
- Starting from the truth, truncated d/o 2
- Co-estimation of the low-degree gravity field coefficients with the orbit parameters in the first iteration
- Free solution
- No additional parameters needed



NGA mitigation strategies around Callisto: orbit fit

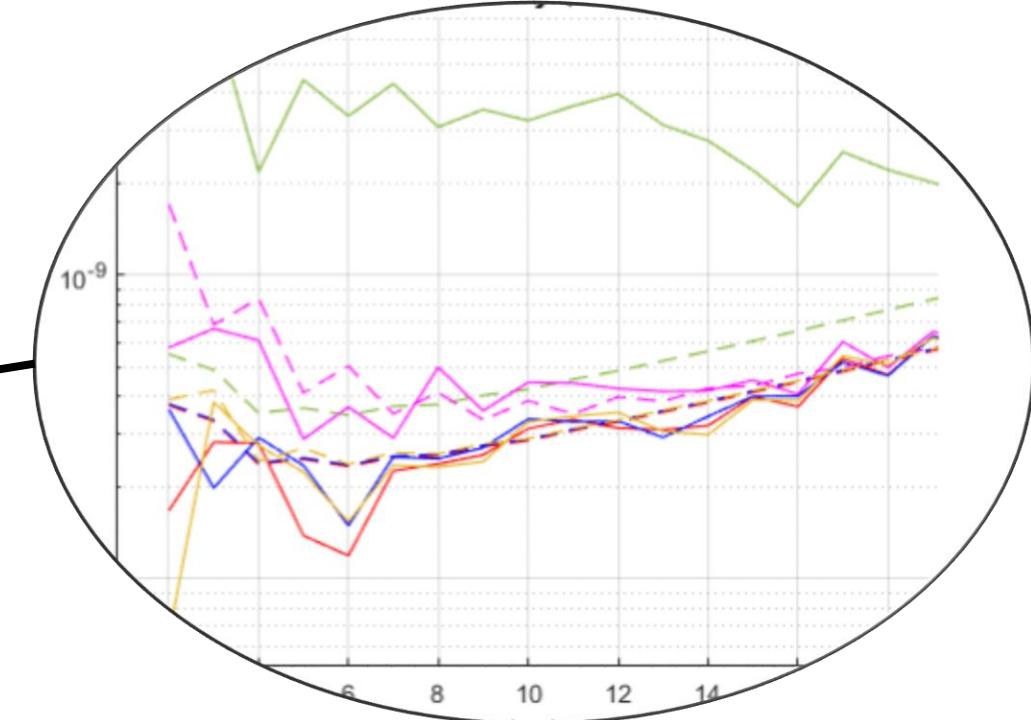
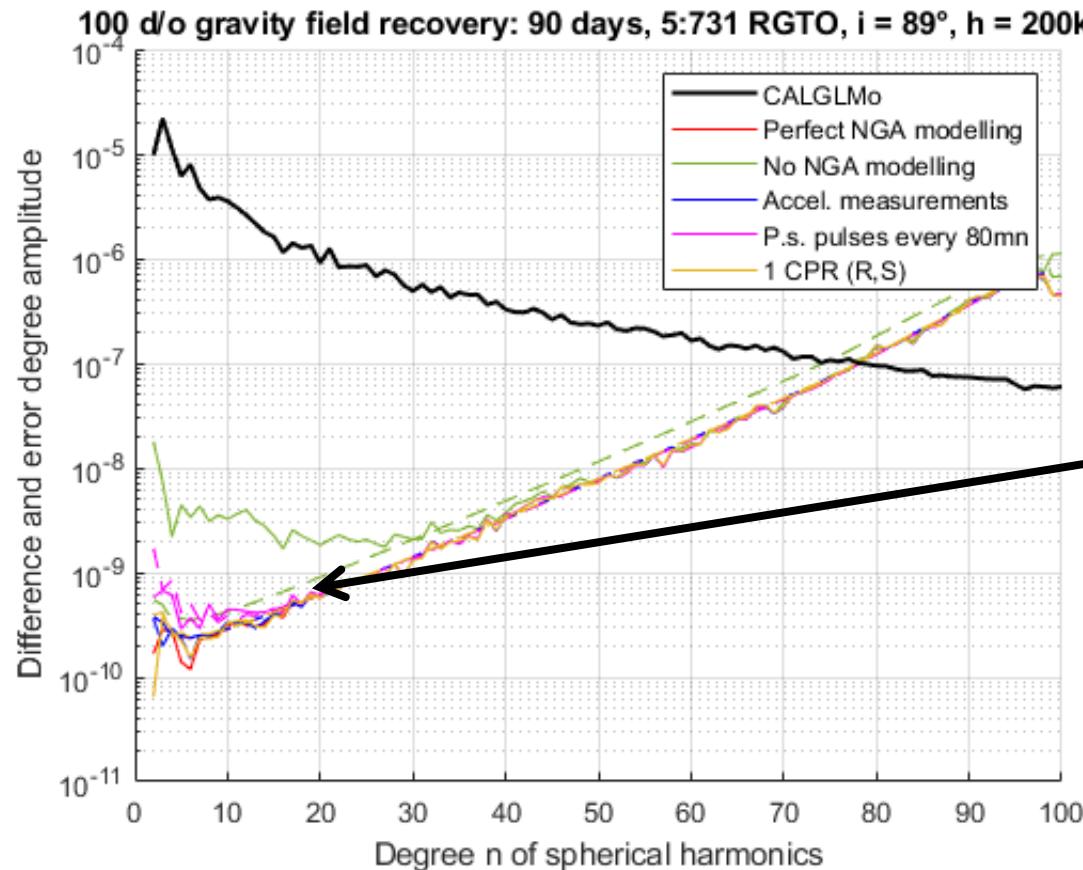
- Different strategies to handle NGA:
 - On-board accelerometer (10^{-8} m/s^2 white noise)
 - Pseudo-stochastic pulses (every 80mn = $\frac{1}{2}$ orbital period)
 - 1 Cycle Per Revolution acceleration in radial and along-track directions

Average orbit distance between the simulated reference orbit, and the final estimated orbit for each 25h arcs.



NGA mitigation strategies around Callisto: gravity field recovery

- Different strategies to handle NGA:
 - On-board accelerometer (10^{-8} m/s 2 white noise)
 - Pseudo-stochastic pulses (every 80mn = $\frac{1}{2}$ orbital period)
 - 1 Cycle Per Revolution acceleration in radial and along-track directions



Summary

- Influence of several orbital characteristic on gravity field recovery
- Comparison of several strategies using a very poor knowledge a priori gravity field
- Non-gravitational acceleration mitigation

Outlook

- Estimation of rotation parameters
- Simulation of altimetry measurements
- Altimetry crossover analysis to constrain orbit and geodetic parameters

Thank you for your attention!